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IN - FLIGHT CHECKS Air-Driven Generator Checks

Vol. 1 09–02–1 REV 41, Apr 17/01

GENERAL

The following procedures are recommended to operationally check the air-driven generator (ADG) in flight using the manual deployment system.



- The presence of foreign objects in the ADG bay could cause damage to the ADG blades during deployment. Before flight, the ADG bay must be inspected to ensure that foreign objects are not present.
- On airplanes 5301 to 5407, when deployed, the ADG is maintained in position by an ejection spring and airloads. After landing, high deceleration could cause the ADG to swing forward with resulting damage to the ADG blades and adjacent fuselage panels. When landing with the ADG deployed, plan to use maximum landing roll-out available to minimize deceleration.

NOTE

Once deployed, the ADG cannot be stowed during flight. Therefore, this check should be performed at the end of a test flight. Normal electrical power must be restored before final landing approach is commenced.

If icing conditions are encountered while the ADG is the only source of electrical power, the WING anti-ice system must be operated only in the standby (STBY) mode.

MANUAL DEPLOYMENT OPERATIONAL CHECK

Before Flight:

closed.

- CL-604 Operating Manual
 PSP 604-6

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MANUAL DEPLOYMENT OPERATIONAL CHECK (CONT'D)

In-	-Flight Procedures:
1.	BATTERY MASTER Check ON
2.	APU Start
3.	APU GEN OFF
4.	VHF Comm, VHF Nav, ADF and transponder Set to #1
	NOTE
	The No. 2 VHF Comm, VHF Nav, ADF and transponder become inoperative when the main generators (GEN 1 and GEN 2) are turned off in the steps to follow.
5.	HYDRAULIC, 1 and 2 switches ON
6.	Airspeed Maintain 180 KIAS
7.	Flaps Set to 20°
8.	Altitude Maintain 10,000 feet or below
9.	ADG DEPLOY CONT - AUTO circuit breaker (2N6) Open
10.	Landing gear lever Select to DN
11.	HYDRAULIC, 3A and 3B switches OFF
	 Bleed off No. 3 system pressure and check that HYD 3 LO PRESS caution message comes on.
12.	AC ELECTRICAL synoptic page Select
13.	ADG manual deploy handle Pull
	Pull the handle for one (1) second to ensure proper opening of the uplock.
	ADG manual deploy handle Stow
15.	AC ELECTRICAL page
	Verify that within six (6) seconds, the following indications occur: ADC vertex = 415 + 51400
	 ADG voltage
	(± 20 Hz on airplanes 5301 to 5407)
	AC ESS BUS Powered
	 AC essential bus temporarily not powered (AC ESS BUS caution message on EICAS comes on briefly, then out).
	NOTE
	On airplanes 5408 to 5572 not incorporating Service Bulletin 604–24–013 operating on emergency power only, the AC ESS BUS caution message may remain on or be intermittent.
	ADG output Green
16.	DC ELECTRICAL synoptic page Check
	Verify the following indications:
	ESS TRU 1, DC ESS BUS and BATT BUS Boward
	and BATT BUS Powered

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MANUAL DEPLOYMENT OPERATIONAL CHECK (CONT'D)

In-Flight Procedures (Cont'd):

17. HYDRAULIC synoptic page Check

• That system pressure shows 3,000 ± 200 psi.



If the ADG fails to deliver the required electrical power after the GEN 1 and GEN 2 switches are set to OFF, the GEN 1 and GEN 2 switches must be reset immediately to ON and normal supply to the essential busses restored by pressing the PWR TXFR OVERRIDE switch, after confirming that the ADG manual deploy handle is stowed.

	after confirming that the ADG manual deploy handle is stowed.
18.	GEN 1 OFF
	Check that GEN 1 OFF caution message comes on.
19.	GEN 2 OFF
	Check that GEN 2 OFF caution message comes on.
20.	Primary display Check
	 That EMER PWR ONLY warning message comes on,
	MACH TRIM caution message comes on, and
	• All other indications are normal, except VIB=0.0.
21.	Pilot's, Copilot's and
	third crewmember's intercoms
22.	STAB TRIM, CH 2 switch/light Check engaged
	STAB TRIM CH 1 INOP status message on.
23.	YAW DAMPER, YD 1 switch/light Check engaged
	YD 1 INOP status message out.

24. Pitch trim Operate

· To check that stabilizer trim operates normally.

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Vol. 1 09–02–4 REV 45, Mar 13/02

MANUAL DEPLOYMENT OPERATIONAL CHECK (CONT'D)

In-Flight Procedures (Cont'd):

- 25. AC and DC essential busses Check
 - Verify that power is available on each bus. It is sufficient to confirm
 the normal operation of one of the services powered by the bus.
 The following is a list of the essential busses and the services
 appropriate for this check:

A. DC Essential Bus:

- VHF Nav 1 operational.
- ADF 1 operational.
- Transponder 1 operational.
- Pilot's MFD and PFD with complete displays.
- Wing anti-ice, STBY (Check for ITT variation).

B. AC Essential Bus:

- · ICE DET 1 FAIL status message out.
- ICE DET TEST: Select to DET and verify that L AOA, L & STBY PITOT, L STATIC and L WINDOW HEAT caution messages out; then

LH WHSLD switch RESET

NOTE

L AOA HEAT, PITOT BASE HEAT, L PITOT HEAT and STBY PITOT HEAT caution messages may come on during ADG deployment. If this occurs, reset circuit breaker 4A11.

C. Battery Bus:

- RAM AIR SOV operational.
- · Copilot's clock operational.
- Standby attitude indicator and altimeter/indicated airspeed indicators operational.
- INTEG instruments panel lights operational.
- 26. Hydraulic system No. 3 pressure Stabilized at 3,000 ±200 psi
- 27. Landing gear lever Cycle; UP then DN
 - Check that hydraulic No. 3 system pressure does not go below 2,200 psi.
- - Slow down to 130 KIAS or slowest practical safe speed and confirm the ADG output:
 - ADG voltage 115 ± 5 VAC
 - Frequency 340 Hz minimum
- 29. Airspeed As required
 - Resume normal airspeed for phase of flight, in accordance with the operating limitations as established in Chapter 2.

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09-02-5 Vol. 1 REV 37, Oct 12/99

MANUAL DEPLOYMENT OPERATIONAL CHECK (CONT'D)

In-Flight Procedures (Cont'd):

- 30. GEN 1 and GEN 2 ON
 - Check that GEN 1 OFF and GEN 2 OFF caution messages go out;
 - Check that EMER PWR ONLY warning message goes out.
- 31. ADG, PWR TXFR OVERRIDE switch Press
 - To restore normal power supply.
 - · Check that all indications return to normal.
- 32. ADG DEPLOY CONT AUTO circuit breaker (2N6) Close

33. HYDRAULIC, 3A and 3B switches As required

• The hydraulic 3A switch is normally set to ON and 3B switch is normally set to AUTO.



On airplanes 5301 to 5407:

On landing, with the ADG deployed, use gentle deceleration to preclude the ADG from swinging forward and damaging the fuselage.